HK 36 TTC AFM
HK 36 TTS AFM
HK 36 TTC ECO AFM



Temporary Revision
Engine cool down
before shutdown

TEMPORARY REVISION TR-MÄM-36-339/b

supersedes TR-MÄM-36-339/a

Engine cool down before shutdown

This Temporary Revision TR-MÄM-36-339/b is approved in conjunction with the Mandatory Design Change Advisory MÄM 36-339/b and is valid in conjunction with the latest revision of the HK 36 TTC, HK 36 TTS and HK 36 TTC ECO - Airplane Flight Manual until this temporary revision has been incorporated into the Airplane Flight Manual.

The limitations and information contained herein either supplement or, in the case of conflict, override those in the Airplane Flight Manual.

The technical information contained in this document has been approved under the authority of DOA No. EASA.21J.052.

| Doc. No. | Section | Affected Pages | |
|-----------|---------|----------------|--|
| 3.01.20-E | 4 | 4-18a, 4-26a | |
| 3.01.15-E | 4 | 4-18a, 4-26a | |
| 3.01.25-E | 4 | 4-19a, 4-29a | |

Instruction

- -Print this document on yellow paper (single-sided).
- -Insert this cover page as the first page of the AFM.
- Insert the other pages of this Temporary Revision in front of the corresponding AFM pages.

| · | Doc. No. 3.01.20-E | | | |
|---|--------------------|-----------------|-------------|------------|
| | Doc. No. 3.01.15-E | TR-MÄM-36-339/b | 25-May-2012 | Cover Page |
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4.5.3 (INCLUDING IN-FLIGHT ENGINE STOP/START PROCEDURES)

The CAUTION in section 4.5.3 is amended to read:

CAUTION

In order to avoid overheating the bearings in the turbocharger, the engine must be cooled down at idle power for a minimum of two minutes before being turned off.

| Doc. No. 3.01.20-E | | | 4 - 18a |
|--------------------|-----------------|-------------|---------|
| Doc. No. 3.01.15-E | TR-MÄM-36-339/b | 25-May-2012 | 4 - 18a |
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4.5.10 ENGINE SHUT-DOWN

The CAUTION in section 4.5.10 is amended to read:

CAUTION

In order to avoid overheating the bearings in the turbocharger, the engine must be cooled down at idle power for a minimum of two minutes before being turned off. This should be observed especially after engine test running. Sufficient cooling usually occurs through landing approach and subsequent taxiing.

| Doc. No. 3.01.20-E | | | 4 - 26a |
|--------------------|-----------------|-------------|---------|
| Doc. No. 3.01.15-E | TR-MÄM-36-339/b | 25-May-2012 | 4 - 26a |
| Doc. No. 3.01.25-E | | | 4 - 29a |